

7 FIL COPY

2

# David Taylor Research Center

Bethesda, MD 20084-5000

AD-A231 706

## 60th Shock and Vibration Symposium

### Volume III

Hosted by the  
David Taylor Research Center  
Underwater Explosions Research Division  
Portsmouth, Virginia

Proceedings of a conference sponsored by the  
Department of Defense, the National Aeronautics and  
Space Administration, and the Department of Energy  
held in Virginia Beach, Virginia  
November 14-16, 1989

DTIC  
ELECTE  
FEB 07 1991  
S B D



Approved for public release; distribution is unlimited.

91 2 04 189

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.				
1. AGENCY USE ONLY (Leave blank)		2. REPORT DATE November 1989	3. REPORT TYPE AND DATES COVERED Proceedings of Meeting	
4. TITLE AND SUBTITLE The Shock Test Facility: An Explosive-Driven, Water-Filled Conical Shock Tube			5. FUNDING NUMBERS Work Unit #59-0584-0-0 Assession #DN780-137 P.E. #63504N Task S0223 & S1704	
6. AUTHOR(S) J. F. Zalesak and L. B. Poche'				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Methods Section, Measurements Branch, Underwater Sound Reference Detachment Naval Research Laboratory P.O. Box 568337, Orlando, FL 32856-8337			8. PERFORMING ORGANIZATION REPORT NUMBER  N/A	
9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) Naval Sea Systems Command Washington, DC 20362-5101			10. SPONSORING / MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES This article appeared in the David Taylor Research Center Proceedings of the November 1989 60th Shock and Vibration Symposium, Vol. III, pp 73-76				
12a. DISTRIBUTION / AVAILABILITY STATEMENT  Approved for public release; distribution unlimited			12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) Shock damage to hull-mounted sonar transducers and related components caused by exploding ordnance is of great concern to the Navy. Consequently, sonar transducers are explosive shock tested at the West Coast Shock Facility, Hunters Point, San Francisco. A conical shock tube is being developed at NRL-USRD as an inexpensive alternative to the WCSF test. The shock tube is a water-filled, conical-bore tube about 8-m long with a 50-cm ID at the large end. Operating parameters are being determined by using an existing small conical shock tube. The conical geometry is chosen because it represents a small solid angle segment of the spherically expanding field in open water. The charge is reduced by the fraction of a sphere represented by the solid angle. The transducer under test is mounted to a piston in the cylindrical chamber at the large end of the tube. The resulting shock wave propagates down the conical tube, strikes the transducer, and produces the pressure shock. The expanding gas bubble then accelerates the piston along its tube, resulting in inertial shock. Extensive testing was performed in a prototype tube to determine proper breech design and amount of explosive needed to reproduce levels in tests at WCSF.				
14. SUBJECT TERMS Shock Test Explosive Shock Test Shock Tube			15. NUMBER OF PAGES 4	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT SAR	

## GENERAL INSTRUCTIONS FOR COMPLETING SF 298

The Report Documentation Page (RDP) is used in announcing and cataloging reports. It is important that this information be consistent with the rest of the report, particularly the cover and title page. Instructions for filling in each block of the form follow. It is important to **stay within the lines** to meet **optical scanning requirements**.

**Block 1. Agency Use Only (Leave blank).**

**Block 2. Report Date.** Full publication date including day, month, and year, if available (e.g. 1 Jan 88). Must cite at least the year.

**Block 3. Type of Report and Dates Covered.** State whether report is interim, final, etc. If applicable, enter inclusive report dates (e.g. 10 Jun 87 - 30 Jun 88).

**Block 4. Title and Subtitle.** A title is taken from the part of the report that provides the most meaningful and complete information. When a report is prepared in more than one volume, repeat the primary title, add volume number, and include subtitle for the specific volume. On classified documents enter the title classification in parentheses.

**Block 5. Funding Numbers.** To include contract and grant numbers; may include program element number(s), project number(s), task number(s), and work unit number(s). Use the following labels:

<b>C</b> - Contract	<b>PR</b> - Project
<b>G</b> - Grant	<b>TA</b> - Task
<b>PE</b> - Program Element	<b>WU</b> - Work Unit Accession No.

**Block 6. Author(s).** Name(s) of person(s) responsible for writing the report, performing the research, or credited with the content of the report. If editor or compiler, this should follow the name(s).

**Block 7. Performing Organization Name(s) and Address(es).** Self-explanatory.

**Block 8. Performing Organization Report Number.** Enter the unique alphanumeric report number(s) assigned by the organization performing the report.

**Block 9. Sponsoring/Monitoring Agency Name(s) and Address(es).** Self-explanatory.

**Block 10. Sponsoring/Monitoring Agency Report Number.** (If known)

**Block 11. Supplementary Notes.** Enter information not included elsewhere such as: Prepared in cooperation with...; Trans. of...; To be published in... When a report is revised, include a statement whether the new report supersedes or supplements the older report.

**Block 12a. Distribution/Availability Statement.** Denotes public availability or limitations. Cite any availability to the public. Enter additional limitations or special markings in all capitals (e.g. NOFORN, REL, ITAR).

**DOD** - See DoDD 5230.24, "Distribution Statements on Technical Documents."

**DOE** - See authorities.

**NASA** - See Handbook NHB 2200.2.

**NTIS** - Leave blank.

**Block 12b. Distribution Code.**

**DOD** - Leave blank.

**DOE** - Enter DOE distribution categories from the Standard Distribution for Unclassified Scientific and Technical Reports.

**NASA** - Leave blank.

**NTIS** - Leave blank.

**Block 13. Abstract.** Include a brief (*Maximum 200 words*) factual summary of the most significant information contained in the report.

**Block 14. Subject Terms.** Keywords or phrases identifying major subjects in the report.

**Block 15. Number of Pages.** Enter the total number of pages.

**Block 16. Price Code.** Enter appropriate price code (*NTIS only*).

**Blocks 17. - 19. Security Classifications.** Self-explanatory. Enter U.S. Security Classification in accordance with U.S. Security Regulations (i.e., UNCLASSIFIED). If form contains classified information, stamp classification on the top and bottom of the page.

**Block 20. Limitation of Abstract.** This block must be completed to assign a limitation to the abstract. Enter either UL (unlimited) or SAR (same as report). An entry in this block is necessary if the abstract is to be limited. If blank, the abstract is assumed to be unlimited.

## SYMPOSIUM MANAGEMENT

Host Representative  
Robert E. Fuss, DTRC/UERD

Contract Management  
J. Robert Kretzel, DTRC/UERD

## SYMPOSIUM MANAGER

Henry C. Pusey  
4193 Sudley Road  
Haymarket, VA 22069

## REGISTRATION MANAGER

Sallie C. Pusey

## 60th SYMPOSIUM PROGRAM COMMITTEE

James W. Daniel  
U.S. Army Missile Command  
AMSML-RD-TE-C  
Redstone Arsenal, AL 35898

Robert E. Fuss  
DTRC/UERD  
Norfolk Naval Shipyard  
Bldg. 369  
Portsmouth, VA 23709-5000

Richard S. Pappa  
NASA/Langley Research Center  
MS 230  
Hampton, VA 23665

Jerome Pearson  
U.S. Air Force  
AFWAL/FIBG  
Wright-Patterson AFB, OH 45433

David O. Smallwood  
Sandia National Laboratories  
Division 7544  
P.O. Box 5800  
Albuquerque, NM 87185

Nien-Tsyr Tsai  
Defense Nuclear Agency  
SPSD  
6801 Telegraph Road  
Alexandria, VA 22310

William W. Wassmann  
Naval Surface Warfare Center  
Code U10  
Silver Spring, MD 20903-5000

## COMPILATION OF PROCEEDINGS

Henry C. Pusey  
Sallie C. Pusey

## PRODUCTION OF PROCEEDINGS

Vibration Institute

Accession For	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By _____	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A-1	20

# TABLE OF CONTENTS

## Papers Appearing in Volume III

<b>DYNAMIC TESTING</b> .....	1
Simulating the Damage Potential of Airblast Shock Loads by Abbreviated Shock Pulses J.F. Polk, F.B. Safford, and A.M. See .....	3
Experimental Techniques for Large Amplitude Vibration of Curved Plates C.F. Ng .....	19
Acoustic and Sonic Fatigue Environment of the C-141A Aircraft with Universal Nose Modification L.L. Shaw .....	35
Location of Interior Impact Source in Randomly Excited Structures T.L. Paez, T. Ernest, and S. McBride .....	51
VIBRAFUGE - Combined Vibration and Centrifuge Testing J.D. Rogers, F. Cericola, J.W. Doggett, and M.L. Young .....	63
The Shock Test Facility: An Explosive-Driven, Water-Filled Conical Shock Tube J.F. Zalesak and L.B. Poche, Jr. ....	73
<b>SHOCK CHARACTERIZATION</b> .....	77
Use and Misuse of Shock Spectra R.L. Bort .....	79
High-Resolution Shock-Reponse Spectrum A.R. Gondeck .....	87
Characterizing Transient Vibrations Using Band Limited Moments D.O. Smallwood .....	93
Comparison of Shock Severity Measures T.J. Baca .....	113
Damping and Record Length Effects on Shock Spectra and Shock Design Values P.F. Cunniff and G.J. O'Hara .....	129
Parameter Specification for Shaker Shock Waveform Synthesis-Damped Sines and Wavelets D.B. Nelson .....	151
<b>ISOLATION AND DAMPING</b> .....	195
An Accurate Temperature Shift Function and a New Approach to Modeling Complex Modulus L. Rogers .....	197
Development of a Novel Vibration Isolation System for Spacecraft Structures D.P. Colvin and A.L. Patra .....	217
Feasibility of Treadmill Isolation for the Space Station Freedom Microgravity Environment C.E. Larsen and T.T. Cao .....	227
Active Isolation Mounts P. Lewis and H.M. Chen .....	241
Precise Positioning Shock Isolators D.P. Taylor and D.A. Lee .....	247
TOMAHAWK Liquid Spring Parameter Determination Through Mathematical Modeling of Test Data J. Leifer and N. Bedewi .....	277
Numerical Simulation of a Multi-Compartmented Gun Muffler and Comparison with Experiment K.S. Fansler, C.H. Cooke, W.G. Thompson, and D.H. Lyon .....	293
The Optimum Design of the Lateral Shock Isolation System of a Canisterized Missile in a Silo Subjected to Lateral Ground Shock D. Orne and R.M. Lakey .....	313

# **THE SHOCK TEST FACILITY AN EXPLOSIVE-DRIVEN, WATER-FILLED CONICAL SHOCK TUBE**

**J.F. Zalesak and L. B. Poché, Jr.  
Naval Research Laboratory  
Underwater Sound Reference Detachment  
Orlando, FL 32806**

Shock damage to hull-mounted sonar transducers and related components caused by exploding ordnance is of great concern to the Navy. Consequently, sonar transducers are required to survive an explosive shock test presently performed at the West Coast Shock Facility (WCSF) at Hunter's Point, San Francisco, CA. A conical shock tube is an inexpensive alternative to the WCSF test. Such a shock test facility being developed at NRL-USRD. The shock tube is a water-filled, conical-bore tube about 8 m long with an inner diameter of 50 cm at the large end.

A small conical shock tube constructed previously is being used to determine operating parameters required for the design of a full scale shock tube. The conical geometry has been chosen because it represents a small solid angle segment of the spherically expanding field in open water. The charge required to produce a specified shock-wave pressure in open water is reduced by the fraction of a sphere represented by the solid angle. The transducer under test is mounted to a piston located in a cylindrical chamber at the large end of the shock tube. The shock wave resulting from the explosive propagates down the conical tube and strikes the transducer, producing the pressure shock. The expanding gas bubble from the explosive then accelerates the piston along its tube, resulting in an inertial shock. Extensive testing was performed in the prototype shock tube to establish a proper breech design, and to determine the amount of explosive needed to reproduce the levels of shock in tests at WCSF. A comparison of the pressure-shock waveforms and the resulting displacements obtained at WCSF and in the prototype shock tube is presented. A design of the full-scale shock tube is also shown.

## **INTRODUCTION**

Naval sonar transducers and related components mounted on the hulls of surface ships and submarines are subjected to both inertial shock and acoustic pressure shock waves when an ordnance charge explodes underwater near the vessel. The potential damage and loss of capability is of great concern to the Navy. As a consequence, naval sonar transducers are required to survive an appropriate test performed presently at the West Coast Shock Facility (WCSF) at Hunter's Point, San Francisco, CA. In this test, the transducer is attached to the bottom of a Floating Shock Platform (FSP) and test charges are detonated near the platform in a prescribed series[1]. The transducers are calibrated both before and after the test to evaluate their susceptibility to shock damage. The tests at Hunter's Point are expensive. In addition, environmental restrictions have led to a reduction in the number of tests that can be performed. Shock testing machines are available for testing equipment weighing up to 7,400 lb. However, these shock machines can apply only inertial shock to the equipment under test. The acoustic shock wave cannot be simulated using these machines. Hydrodynamic shock machines are available[2] to apply a pressure pulse to a device under test. However, these machines produce a slowly varying pulse devoid of the high-frequency spectral content of the pressure shock wave, and the inertial component of the shock is absent. An inexpensive alternative to the Hunter's Point test has become increasingly desirable.

A closed-chamber shock test facility is being developed at NRL-USRD to satisfy this need. Our intention has been to develop a shock tube of adequate diameter to allow the most commonly-tested sonar transducers to be evaluated. Small-diameter shock tubes of this type have been previously described[3], but have been used principally as objects of study. This tube design allows exposure to both shock-wave pressure and inertial shock in a single test by mounting the transducer on a piston which moves in a reaction chamber. The shock-tube dimensions and charge weight are chosen to closely simulate both the peak shock-wave pressure and the FSP's inertial motion during Shot 4 of the heavyweight shock test schedule of reference 1.

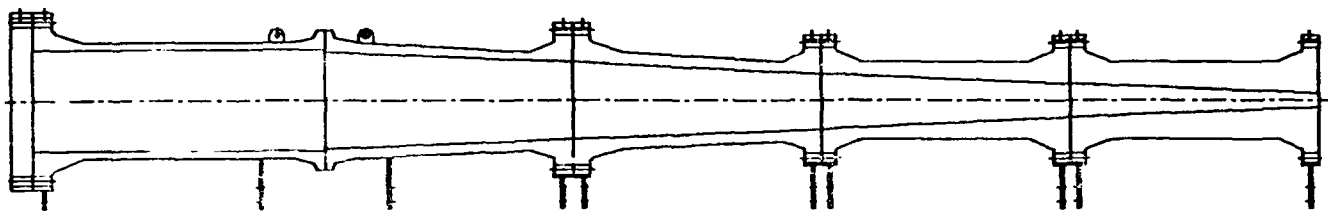


Fig. 1. Full-scale shock tube design.

The shock tube is a water-filled, conical-bore tube about 8-m long with an inner diameter of 50 cm at the large, or muzzle, end (Fig. 1). Fabrication cost effectively limits the chamber size to 50 cm, which is sufficient for mounting a (size of transducer) at angles of up to 50 degrees. The environmental constraints of operating such a shock tube are minuscule compared to the practice of detonating 60 lb of high explosive in an urban estuary, as must be done at WCSF. Shock tube tests can be performed with a frequency limited only by the mechanics of opening, loading and filling the tube. In operation, the tube is mounted horizontally and a small explosive charge (5-20 gm TNT equiv.) is detonated in the breech, or small diameter end of the tube. This produces the pressure shock wave, which quickly travels down the tube. The transducer under test is mounted on the face of a piston free to slide in a cylindrical chamber, and forms the termination of the tube. After the shock wave strikes the transducer, the expanding gas bubble from the explosion accelerates the piston into the chamber, applying the inertial shock to the transducer.

### EXPERIMENTS

A small (15-cm diameter, 3-m length) conical shock tube (Fig. 2) which was built and used in previous experiments is being employed to determine various operating parameters required in the design of the full-scale shock tube.

Filler[3] showed shock wave pressure records that have high-frequency energy of broad spectral range superimposed on the exponential pressure decay of the shock wave. This feature seems to be a characteristic of the shock wave generated in a conical tube with thick steel walls. A potential secondary pressure pulse resulting from the reflection of the shock wave from the piston surface may be absorbed sufficiently by covering the piston surface with a water-soaked cypress wood cap of the proper grain orientation[4]. A typical sample of a shock-wave pressure measurement made in the closed tube and recorded with a 1/4-in diam. tourmaline disk gage is shown in Fig. 3. The gage output was recorded on a digital waveform recorder. The estimated peak pressure obtained for this shot, which was generated from an electric blasting cap of 0.65-gm TNT equivalent, is 2380 psi. The high-frequency energy has a strong

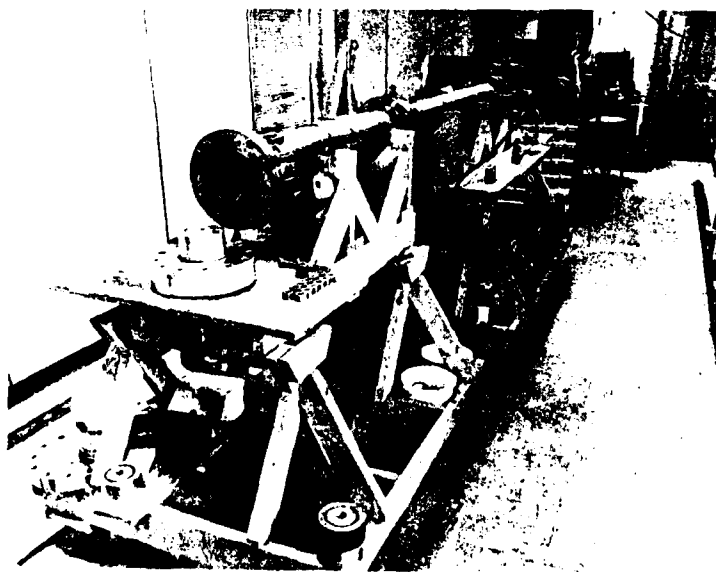


Fig. 2. 15-cm diam. shock tube.

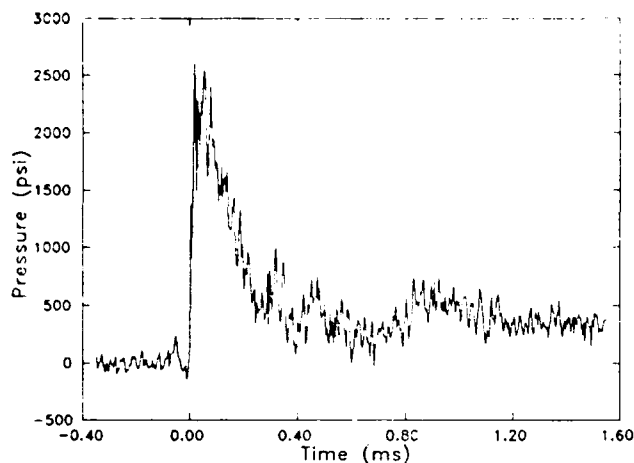


Fig. 3. Shock-wave pressure measurement from 15 cm tube.

a half-sine pulse of approximately 600-ms duration and 16-in peak displacement. The vertical kickoff velocity for the FSP has been computed as 10.7 fps. This agrees quite well with the reported waveform.

Certain problems that arose during the experimentation have solutions that, at best, consist of compromises. The problem of initiating the charge in a safe and convenient manner still remains. Openings at the breech end of the tube made to admit firing wires erode rapidly and also form stress concentrations. At present, we are running firing lines down the length of the tube, where they can be brought out on standard connectors, but must be replaced for every shot. The breech is designed with a removable block, which is in the form of a steel cylinder with a cavity on one face to hold the explosive. An extensive study was done to find the best design and the most cost-effective material for this block. Titanium proved the most durable material, but AISI 1018 steel is the choice when the breech block can be replaced

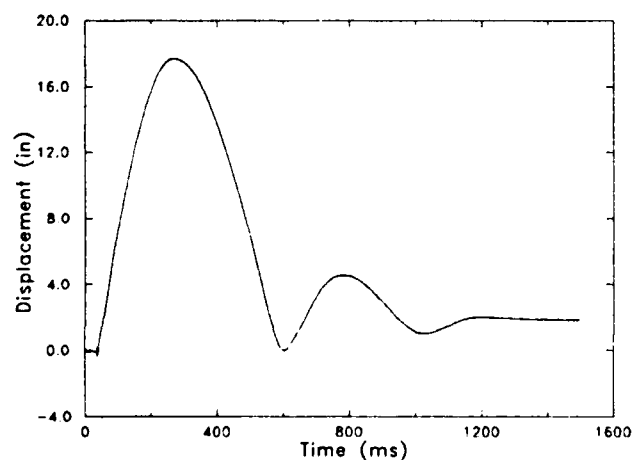


Fig. 5. Reaction block motion in 15 cm tube.

visual effect on the appearance of the pressure-wave signature, but does not contribute much to the total energy in the spectrum. This signature may be compared with one recorded similarly at WCSF at a 30-ft standoff test (Shot 3 of MIL-S-901D), seen in Fig. 4.

The inertial shock motion of our reaction chamber piston was recorded by attaching the core of a 20-in linear variable differential transformer (LVDT) to the rear of the sliding piston. It was assumed that the mass of the piston and the water in the tube would be controlled by the spring rate of the gas bubble generated in the explosion. The period of this oscillator was adjusted by adding mass to the piston. Fig. 5 shows the time history of the piston displacement when the tube is driven by a charge consisting of an E-1A blasting cap and 0.7 gm of DuPont Detaprime GA. The piston was loaded with approximately 200 lb of lead. The inertial shock motion of the WCSF FSP has been reported previously[5], and is taken to be

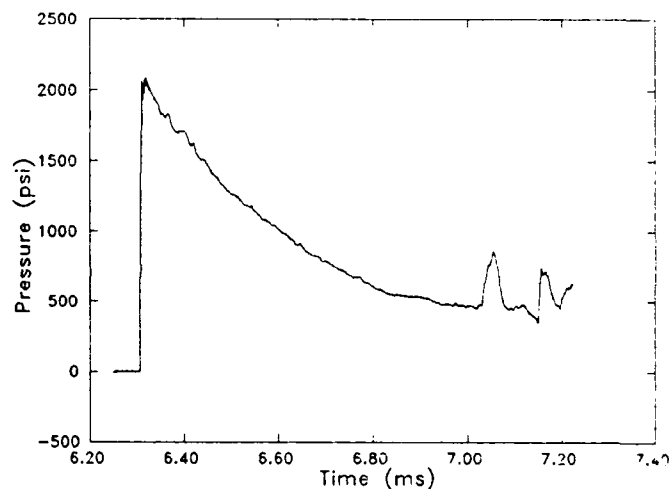


Fig. 4. Shock wave pressure at FSP.

frequently. Another problem is the vibration that couples into the steel tube that then reradiates as sound into the water. This effect is what causes the shock wave pressure signature to appear "noisy". Normally, we estimate the peak pressure level by the method described by Cole[6], but the additional "noise" signal on the decay portion of the curve makes the extrapolation procedure less exact. We were able to reduce the "steel" portion of the signal somewhat by mechanically isolating the tube flanges and breech block with elastomeric gaskets, but this greatly increased the frequency of blowouts and leaks. Finally, the problem of corrosion strongly affects the choice of material for the tube itself. Nickel-aluminum bronze with integrally-cast flanges is the material of choice from the point of view of strength and corrosion resistance, however the cost is prohibitive. Using welded-on steel flanges lowers the cost to an acceptable range, but creates unsolvable manufacturing difficulties. The best compromise, considering all factors is stainless steel 316L with integrally-cast flanges.



### CONCLUSION

We have proposed a design for a full-size shock tube facility that will provide either alternative testing or pre-qualifying test results to the WCSF MIL-S-901D compliance. As many as four test cycles per work day may be carried out, and at a fraction of the cost per test.

### REFERENCES

1. MIL-S-901D (NAVY), "Military Specification. Shock Tests, HI (High-Impact); Shipboard Machinery, Equipment, and Systems, Requirements for," 17 March 1989
2. Cameron D. Johnson and Robert J. DeAngelis, "Characteristics of the Hydrodynamic Shock Simulator Developed by the Navy Underwater Sound Laboratory," NUSL Report No. 1029, 3 September 1969
3. W. S. Filler, "Propagation of Shock Waves in a Hydrodynamic Conical Shock Tube," *Phys. Fluids* 7, 664 (1964)
4. L. B. Poché, Jr., L. Dwight Luker and Peter H. Rogers, "Measurement and Analysis of Echolocation Clicks of Free-Swimming Dolphins (*T. truncatus*) in a Tank with Echo-Reducing Wood Lining," NRL Report 8434, May 21, 1981, Appendix
5. E. W. Clements, "Shipboard Shock and Navy Devices for its Simulation," NRL Report 7396, July 14, 1972, p. 117
6. R. H. Cole, Underwater Explosions, pp. 176-177, Princeton Univ. Press, Princeton, 1948, reprinted by Dover Publ., 1965